

ARCHIEF

Lab. v. Scheepsbouwkunde
Technische Hogeschool
Delft

Office of Naval Research
Department of the Navy
Contract Nonr-220(35)

EFFECT OF MODULATION ON THE ONSET OF
THERMAL CONVECTION

by
Giulio Venezian

Division of Engineering and Applied Science
CALIFORNIA INSTITUTE OF TECHNOLOGY
Pasadena, California

Report No. 97-15

March 1968

Office of Naval Research
Department of the Navy
Contract Nonr 220(35)

EFFECT OF MODULATION ON THE ONSET OF
THERMAL CONVECTION

by

Giulio Venezian

Reproduction in whole or in part is permitted
for any purpose of the United States Government

Distribution of this Document is Unlimited
Division of Engineering and Applied Science
California Institute of Technology
Pasadena, California

Abstract

The stability of a horizontal layer of fluid heated from below is examined when, in addition to a steady temperature difference between the walls of the layer, a time-dependent sinusoidal perturbation is applied to the wall temperatures. Only infinitesimal disturbances are considered. The effects of the oscillating temperature field are treated by a perturbation expansion in powers of the amplitude of the applied field. The shift in the critical Rayleigh number is calculated as a function of frequency, and it is found that it is possible to advance or delay the onset of convection by time modulation of the wall temperatures.

I. INTRODUCTION

R. J. Donnelly⁽¹⁾ has reported experiments on the effect of modulation on the stability of the flow between rotating cylinders. In his experiments, fluid was confined in the narrow gap between two cylinders, with the outer cylinder held fixed while the inner cylinder was given an angular speed $\Omega + \Delta\Omega \cos \omega t$. He found that the onset of instability was delayed by the modulation of the angular speed of the inner cylinder. Maximum stability was achieved for $\omega d^2/\nu \simeq 0.27$, and as the frequency was increased far beyond that point, the effect of modulation became negligible. Donnelly interpreted his results as being due to a viscous wave penetrating the fluid and thereby altering the profile from an unstable one to a stable one.

Since the problems of Taylor stability and Bénard stability are very similar, and the latter is simpler to analyze, this paper deals with the thermal analog of Donnelly's experiments. The problem considered is that of determining the onset of convection for a fluid layer heated from below, when in addition to a fixed temperature difference between the walls, an additional perturbation is applied to the wall temperatures, varying sinusoidally in time.

II. STATEMENT OF THE PROBLEM

The problem considered is the following. A fluid layer is confined between two infinite horizontal walls, a distance L apart. A vertical gravity force acts on the fluid. The wall temperatures are externally imposed, and they are

$$T_R + \frac{\Delta T}{2} [1 + \epsilon \cos \omega t] \quad (1)$$

at the lower wall ($z = 0$), and

$$T_R - \frac{\Delta T}{2} [1 - \epsilon \cos(\omega t + \varphi)] \quad (2)$$

at the upper wall ($z = L$). Here ϵ represents a small amplitude.

The fluid is supposed to be essentially incompressible, except insofar as its density changes due to thermal expansion. For small departures from a reference temperature T_R , the density is given by

$$\rho = \rho_R [1 - \alpha(T - T_R)] \quad (3)$$

where α is the coefficient of thermal expansion. The thermal diffusivity κ and the kinematic viscosity ν of the fluid will be regarded as constant, and the Boussinesq⁽²⁾ approximation will be used to describe the motion of the fluid.

For simplicity, "free-free"⁽²⁾ boundary conditions will be applied at the wall, instead of the more physical no-slip conditions. The free-free conditions are that the normal velocity is zero and the tangential stress is zero at the wall. They correspond to a rigid but slippery wall.

The object of the analysis is to determine the critical conditions under which convection can occur.

III. THE HYDROSTATIC CONFIGURATION

A hydrostatic configuration is possible for this system, in which the isothermal surfaces (and hence the isosteric surfaces) are horizontal and therefore parallel to the equipotential surfaces of the vertical gravitational force. Under such conditions a vertical pressure gradient can balance the gravitational force, and the fluid is subject to no net force. The equations which determine the temperature and pressure fields in this case are:

$$-\frac{\partial p_H}{\partial z} = \rho_H g \quad , \quad (4)$$

and

$$\frac{\partial T_H}{\partial t} = \kappa \frac{\partial^2 T_H}{\partial z^2} \quad . \quad (5)$$

Equations (3), (4) and (5) together with the boundary conditions (1) and (2) determined the hydrostatic fields $T_H(z, t)$, $\rho_H(z, t)$ and $p_H(z, t)$.

We shall only need the temperature field T_H , which, since Eq. (5) is linear, consists of the sum of a steady temperature field T_S and an oscillating part ϵT_1 :

$$T_H = T_S(z) + \epsilon T_1(z, t) \quad , \quad (6)$$

where

$$T_S = T_R + \Delta T(L - 2z)/2L \quad , \quad (7)$$

and

$$T_1 = \text{Re}\{[a(\lambda)e^{\lambda z/L} + a(-\lambda)e^{-\lambda z/L}]e^{-i\omega t}\} \quad . \quad (8)$$

In Eq. (8),

$$\lambda = (1 - i) \left(\frac{\omega L^2}{2\kappa} \right)^{\frac{1}{2}} \quad , \quad (9)$$

and

$$a(\lambda) = \frac{\Delta T}{2} \frac{e^{-i\varphi} e^{-\lambda}}{e^{\lambda} - e^{-\lambda}} \quad . \quad (10)$$

IV. EQUATIONS OF MOTION

In the Boussinesq approximation, the equations of motion are

$$\frac{\partial \underline{v}}{\partial t} + \underline{v} \cdot \nabla \underline{v} + \frac{1}{\rho R} \nabla(p - p_H) = \nu \nabla^2 \underline{v} + g\alpha(T - T_H)\underline{k} \quad , \quad (11)$$

$$\nabla \cdot \underline{v} = 0 \quad , \quad (12)$$

and

$$\frac{\partial T}{\partial t} + \underline{y} \cdot \nabla T = \kappa \nabla^2 T \quad , \quad (13)$$

where \underline{k} is the unit vector in the vertical direction, and $\underline{y} = (u, v, w)$ is the fluid velocity.

Let

$$\theta = T - T_H \quad , \quad (14)$$

then, retaining only linear terms in \underline{y} and θ , the equations of motion are

$$\frac{1}{\sigma} \frac{\partial \underline{y}'}{\partial t'} + \nabla' p' = \nabla'^2 \underline{y}' + R \theta' \underline{k} \quad , \quad (15)$$

$$\nabla' \cdot \underline{y}' = 0 \quad , \quad (16)$$

and

$$\frac{\partial \theta'}{\partial t'} + w' \frac{\partial T'_H}{\partial z'} = \nabla'^2 \theta' \quad . \quad (17)$$

Here, the variables have been non-dimensionalized as follows:

$$\underline{x}' = \underline{r}/L \quad , \quad t' = \kappa t/L^2 \quad , \quad T' = T/\Delta T \quad ,$$

$$\underline{y}' = \frac{L}{\kappa} \underline{y} \quad , \quad p' = Lp/\rho_R \kappa^2 \quad .$$

The two dimensionless groups which appear are the Prandtl number, $\sigma = \nu/\kappa$, and the Rayleigh number, $R = g\alpha\Delta TL^3/\kappa\nu$.

From this point on we shall drop the primes, with the understanding that, unless otherwise stated, the quantities are in their non-dimensional form.

The boundary conditions at $z = 0$ and $z = 1$ are:

$$w = 0 \quad (\text{rigid wall}),$$

$$\frac{\partial^2 w}{\partial z^2} = 0 \quad (\text{slippery walls}),$$

and

$$\theta = 0 \quad (\text{externally fixed temperature}).$$

We are interested in non-zero solutions to Eqs. (15) - (17) subject to these boundary conditions.

It is convenient to express the entire problem in terms of w . This is accomplished by taking the curl of Eq. (15) twice. The z component of the resulting equation involves only w and θ :

$$\left(\frac{1}{\sigma} \frac{\partial}{\partial t} - \nabla^2 \right) \nabla^2 w = R \nabla_1^2 \theta, \quad (18)$$

where

$$\nabla_1^2 \equiv \partial^2 / \partial x^2 + \partial^2 / \partial y^2.$$

Equations (17) and (18) can then be combined to obtain

$$\left(\frac{1}{\sigma} \frac{\partial}{\partial t} - \nabla^2 \right) \left(\frac{\partial}{\partial t} - \nabla^2 \right) \nabla^2 w = -R \frac{\partial T_o}{\partial z} \nabla_1^2 w. \quad (19)$$

The boundary conditions can also be expressed in terms of w by making use of Eq. (18), which requires $\partial^4 w / \partial z^4 = 0$ if w and θ are zero. Thus, Eq. (19) has to be solved subject to the homogeneous conditions

$$w = \partial^2 w / \partial z^2 = \partial^4 w / \partial z^4 = 0 \quad \text{at } z = 0, 1. \quad (20)$$

The temperature gradient appearing in Eq. (19) can be obtained from the expressions derived in Section III:

$$\begin{aligned} \frac{\partial T_o}{\partial z} &= -1 + \epsilon \operatorname{Re} \{ [A(\lambda) e^{\lambda z} + A(-\lambda) e^{-\lambda z}] e^{-i\omega t} \}, \\ &= -1 + \epsilon f \end{aligned} \quad (21)$$

where

$$A(\lambda) = \frac{\lambda}{2} \frac{e^{-i\varphi} - e^{-\lambda}}{e^{\lambda} - e^{-\lambda}}$$

The horizontal dependence of w is factorable in this problem, and we shall study only solutions with a single wavenumber α , such that

$$\nabla_1^2 w = -\alpha^2 w$$

The dependence $e^{i\vec{\alpha} \cdot \vec{r}}$ of w on the horizontal coordinates is to be understood throughout, even though, for the sake of conciseness of notation the exponential factor will be left out.

V. PERTURBATION PROCEDURE

We seek the eigenfunctions w and eigenvalues R of Eqs. (19) and (20) for a temperature profile that departs from the linear profile $\partial T_0 / \partial z = -1$ by quantities of order ϵ . It follows that the eigenfunctions and eigenvalues which obtain in this problem differ from those associated with the standard Benard problem by quantities of order ϵ . Accordingly, we seek an expansion of the form

$$\begin{aligned} w &= w_0 + \epsilon w_1 + \epsilon^2 w_2 + \dots \\ R &= R_0 + \epsilon R_1 + \epsilon^2 R_2 + \dots \end{aligned} \tag{22}$$

This type of expansion was first used in connection with convection problems by Malkus and Veronis⁽³⁾ to consider effects of finite amplitude convection. More recently, a similar expansion was used by Schlüter, Lortz and Busse⁽⁴⁾ to study the stability of finite amplitude convection, and by Ingersoll⁽⁵⁾ to study the effect of superposing a shear flow, among

others. This expansion is in effect a generalization of Rayleigh's perturbation procedure.

If the expansions (22) are substituted into Eq. (19) and the powers of ϵ are separated, the resulting system of equations is

$$\begin{aligned} L w_0 &= 0 \\ L w_1 &= R_{11} \nabla^2 w_0 - R_{of1} \nabla^2 w_0, \\ L w_2 &= R_{11} \nabla^2 w_1 + R_{21} \nabla^2 w_0 - R_{c1} f \nabla^2 w_1 - R_{f1} \nabla^2 w_0, \\ &\dots \end{aligned} \tag{23}$$

where

$$L = \left(\frac{1}{\sigma} \frac{\partial}{\partial t} - \nabla^2 \right) \left(\frac{\partial}{\partial t} - \nabla^2 \right) \nabla^2 - R_o \nabla^2 \tag{24}$$

Each of the w_n 's is required to satisfy the boundary conditions (20).

The function w_0 which starts the whole process is a solution of the problem with $\epsilon = 0$ that is, the classical Bénard problem. The marginally stable solutions for that problem are ⁽²⁾

$$w_0^{(n)} = \sin n\pi z,$$

with corresponding eigenvalues

$$R_o^{(n)} = \frac{(n^2 \pi^2 + \alpha^2)^3}{\alpha^2}.$$

For a fixed value of α the least eigenvalue is

$$R_o = \frac{(\pi^2 + \alpha^2)^3}{\alpha^2}, \tag{25}$$

corresponding to

$$w_0 = \sin \pi z. \tag{26}$$

We shall use these as the starting point of our solution.

The equation for w_1 then reads

$$Lw_1 = -R_1 \alpha^2 \sin \pi z + R_0 \alpha^2 f \sin \pi z \quad . \quad (27)$$

If this equation is to have a solution, the right hand side must be orthogonal to the null space of the operator L . In effect, this solubility condition requires that the time independent part of the right hand side should be orthogonal to $\sin \pi z$. Since f varies sinusoidally in time, the only steady term is $-R_1 \alpha^2 \sin \pi z$, so that R_1 is zero. Indeed, this could have been foreseen because R should be independent of the sign of ϵ , since changing the sign of ϵ merely corresponds to a shift in the time origin by half a period. Since such a shift does not change the problem of stability, it follows that all the odd coefficients R_1, R_3, \dots are zero.

Although Eq. (27) in principle can be solved in closed form, it is more convenient to expand the right hand side in a Fourier series, and thus obtain an expression for w_1 by inverting the operation L term by term. For this, the expansion of $e^{\lambda z}$ in a Fourier series is needed. For subsequent steps in the problem, we require the expansion of $e^{\lambda z} \sin m\pi z$. It is easily determined that

$$\begin{aligned} g_{nm}(\lambda) &= 2 \int_0^1 e^{\lambda z} \sin n\pi z \sin m\pi z \, dz \quad , \\ &= - \frac{4nm\pi^2 \lambda [1 + (-1)^{n+m+1} e^\lambda]}{[\lambda^2 + (n+m)^2 \pi^2][\lambda^2 + (n-m)^2 \pi^2]} \quad , \end{aligned} \quad (28)$$

so that

$$e^{\lambda z} \sin m\pi z = \sum_{n=1}^{\infty} g_{nm} \sin n\pi z \quad . \quad (29)$$

It is convenient to define

$$L(\omega, n) = \frac{\omega^2}{\sigma} (n^2\pi^2 + \alpha^2) + i\omega \left(1 + \frac{1}{\sigma} \right) (n^2\pi^2 + \alpha^2)^2 - (n^2\pi^2 + \alpha^2)^3 + (\pi^2 + \alpha^2)^3 . \quad (30)$$

It follows that

$$L \sin n\pi z e^{-i\omega t} = L(\omega, n) \sin n\pi z e^{-i\omega t} \quad (31)$$

(with the horizontal dependence on $e^{i\mathbf{q}\cdot\mathbf{r}}$ understood).

Equation (27) now reads

$$Lw_1 = R_0 \alpha^2 \operatorname{Re} \left\{ \sum [A(\lambda) g_{n_1}(\lambda) + A(-\lambda) g_{n_1}(-\lambda)] e^{-i\omega t} \sin n\pi z \right\} ,$$

so that

$$w_1 = R_0 \alpha^2 \operatorname{Re} \left\{ \sum \frac{B_n(\lambda)}{L(\omega, n)} e^{-i\omega t} \sin n\pi z \right\} , \quad (32)$$

where

$$B_n(\lambda) = A(\lambda) g_{n_1}(\lambda) + A(-\lambda) g_{n_1}(-\lambda) . \quad (33)$$

A term proportional to $\sin n\pi z$ (the solution to the homogeneous equation) could be added. However, this would merely amount to a renormalization of w , since all the terms proportional to $\sin n\pi z$ could then be regrouped to define a new w_0 , with corresponding new definitions for the other w_n 's.

For this reason, it is convenient to assume from the outset that w_0 is orthogonal to all the other w_n 's.

The equation for w_2 is

$$Lw_2 = -R_2 \alpha^2 w_0 + R_0 \alpha^2 f w_1 \quad (34)$$

We shall not require the solution of this equation, but merely use it to determine R_2 , the first non-zero correction to R . The solubility condition requires that the steady part of the right hand side should be orthogonal to $\sin \pi z$, and therefore

$$R_2 = 2R_0 \int_0^1 \overline{f w_1} \sin \pi z \, dz \quad (35)$$

where the bar denotes a time average.

Now, from Eq. (27)

$$f \sin \pi z = \frac{1}{\alpha^2 R_0} L w_1 \quad ,$$

so that

$$\begin{aligned} \overline{f w_1} \sin \pi z &= \frac{1}{\alpha^2 R_0} \overline{w_1 L w_1} \quad , \\ &= \alpha^2 \frac{R_0}{2} \operatorname{Re} \left\{ \sum \frac{B_n(\lambda)}{L(\omega, n)} \sin n\pi z \sum B_n^*(\lambda) \sin n\pi z \right\} \quad , \end{aligned}$$

and finally

$$\begin{aligned} R_2 &= \alpha^2 \frac{R_0^2}{2} \operatorname{Re} \sum \frac{|B_n(\lambda)|^2}{L(\omega, n)} \quad , \\ R_2 &= \frac{\alpha^2 R_0^2}{4} \sum_{n=1}^{\infty} \frac{|B_n(\lambda)|^2}{|L(\omega, n)|^2} [L(\omega, n) + L^*(\omega, n)] \quad (36) \end{aligned}$$

Equation (34) could now be solved for w_2 if desired, and the procedure continued to evaluate further corrections to w and R . However, we shall stop at this step.

VI. MINIMUM RAYLEIGH NUMBER FOR CONVECTION

The value of R obtained by this procedure is the eigenvalue corresponding to the function w which though oscillating, remains bounded in time. In general R is a function of the horizontal wavenumber α and the amplitude of the perturbation, ϵ . Thus

$$R(\alpha, \epsilon) = R_0(\alpha) + \epsilon^2 R_2(\alpha) + \dots \quad (37)$$

As a function of α there will be a least value R_c of R at say $\alpha = \alpha_c$. This critical value of α occurs when $\partial R / \partial \alpha = 0$, that is when

$$\partial R_0 / \partial \alpha_c + \epsilon^2 \partial R_2 / \partial \alpha_c + \dots = 0 \quad (38)$$

Assume α_c is expanded in powers of ϵ ,

$$\alpha_c = \alpha_0 + \epsilon \alpha_1 + \epsilon^2 \alpha_2 + \dots \quad (39)$$

then Eq. (38) becomes

$$\begin{aligned} \partial R_0 / \partial \alpha_0 + \epsilon (\partial^2 R_0 / \partial \alpha_0^2) \alpha_1 + \epsilon^2 \left[\frac{1}{2} (\partial^3 R_0 / \partial \alpha_0^3) \alpha_1^2 \right. \\ \left. + (\partial^2 R_0 / \partial \alpha_0^2) \alpha_2 + \partial R_2 / \partial \alpha_0 \right] \dots = 0 \quad , \end{aligned}$$

so that

$$\begin{aligned} \partial R_0 / \partial \alpha_0 &= 0 \quad , \\ \alpha_1 &= 0 \quad , \\ \alpha_2 &= - (\partial R_2 / \partial \alpha_0) / (\partial^2 R_0 / \partial \alpha_0^2) \quad . \end{aligned} \quad (40)$$

The first of these expressions gives $\alpha_0^2 = \pi/2$. A similar expansion obtains R_c :

$$\begin{aligned}
 R_c(\epsilon) &= R_{0c} + \epsilon^2 R_{2c} + \epsilon^4 R_{4c} + \dots ; \\
 &= R(\alpha_c, \epsilon) , \\
 &= R_0(\alpha_0) + \epsilon(\partial R_1 / \partial \alpha_0) \alpha_1 + \epsilon^2 \left[\frac{1}{2} (\partial^2 R_0 / \partial \alpha_0^2) \alpha_1^2 \right. \\
 &\quad \left. + (\partial R_0 / \partial \alpha_0) \alpha_2 + R_2(\alpha_0) \right] + \dots \\
 &= R_0(\alpha_0) + \epsilon^2 R_2(\alpha_0) + \dots
 \end{aligned} \tag{41}$$

in view of Eqs. (40). Thus, to order ϵ^2 , R_c is determined by evaluating R_0 and R_2 at $\alpha = \alpha_0$. It is only when one reaches R_4 that α_2 must be taken into account. In the next section, the values of R_{2c} are found for three particular cases.

VII. RESULTS

The values of R_{2c} will be obtained for the following cases: (a) when the oscillating temperature field is symmetric i.e. the plate temperatures are modulated in phase, so $\varphi = 0$; (b) when the field is antisymmetric, corresponding to an out of phase modulation, $\varphi = \pi$; and (c) when only the temperature of the bottom plate is modulated, the upper plate being held at a fixed constant temperature. This case can be recovered from the equations by setting $\varphi = -i\infty$.⁽⁶⁾

In all three cases the expression for $B_n(\lambda)$ simplifies considerably. Let

$$b_n = - \frac{4\pi^2 n \lambda^2}{[\lambda^2 + (n+1)\pi^2][\lambda^2 + (n-1)^2\pi^2]} , \tag{42}$$

then, for case (a)

$$\begin{aligned} B_n &= b_n \text{ if } n \text{ is even} , \\ &= 0 \text{ if } n \text{ is odd} ; \end{aligned}$$

for case (b)

$$\begin{aligned} B_n &= 0 \text{ if } n \text{ is even} , \\ &= b_n \text{ if } n \text{ is odd} ; \end{aligned}$$

and for case (c)

$$B_n = -b_n \text{ for all } n, \text{ (see footnote 6).}$$

The variable λ was defined in Eq. (9), which in terms of the dimensionless frequency reduces to

$$\lambda = (1-i)(\omega/2)^{\frac{1}{2}} ,$$

and thus

$$|b_n|^2 = \frac{16\pi^4 n^2 \omega^2}{[\omega^2 + (n+1)^4 \pi^4][\omega^2 + (n-1)^4 \pi^4]} . \quad (43)$$

We also need an expression for

$$C_n = [L(\omega, n) + L^*(\omega, n)] / 2 |L(\omega, n)|^2$$

evaluated at $\alpha^2 = \pi/2$. This reduces to

$$C_n = \frac{\frac{\omega^2}{\sigma} (n^2+1/2)\pi^2 - (n^2+1/2)^3 \pi^6 + \frac{27\pi^6}{8}}{\left[\frac{\omega^2}{\sigma} (n^2+1/2)\pi^2 - (n^2+1/2)^3 \pi^6 + \frac{27\pi^6}{8} \right]^2 + \omega^2 \left(1 + \frac{1}{\sigma}\right)^2 (n^2+1/2)^4 \pi^8} , \quad (44)$$

and finally

$$R_{2c} = \frac{729}{64} \pi^{10} \sum |b_n|^2 C_n , \quad (45)$$

where the sum extends over even values of n for case (a), odd values for case (b) and all values for case (c). The series defined by Eq. (45) converges rapidly since the terms decrease like $1/n^{12}$.

Numerical results of R_{2c} as a function of ω for various values of σ are exhibited in the accompanying figures.

VIII. DISCUSSION

Some features of the behavior of R_{2c} as a function of ω can be seen by examining the limiting cases for very small or very large values of ω . When ω is very small,

$$C_1 |b_1|^2 \rightarrow 1/[3\sigma(1 + 1/\sigma)^2 \pi^6 / 2] ,$$

while for $n \neq 1$

$$C_n |b_n|^2 \rightarrow - \frac{16 n^2 \omega^2}{(n^2 - 1)^5 (n^4 + \frac{5}{2} n^2 + \frac{13}{4}) \pi^{10}} ,$$

so the general form of R_{2c} near $\omega = 0$ is

$$R_{2c} \rightarrow R_\sigma - \beta \omega^2 ,$$

where

$$R_\sigma = 27\pi^4 / 8\sigma(1 + 1/\sigma)^2 , \quad (46)$$

and β is a constant, which depends only on the case being considered.

In the case of symmetric excitation, the sum extends only over even values, so that

$$R_{2c} \rightarrow - 0.102 \omega^2 ,$$

indicating that the dependence on σ can only appear at reasonably large

values of ω . The effect of modulation in this case is to destabilize the system, with convection occurring at an earlier point than in the unmodulated system. This agrees with the results of Krishnamurti⁽⁷⁾ in her analysis of convection with a slowly varying mean temperature, which corresponds to low frequency symmetric excitation.

In the antisymmetric case

$$R_{2c} \rightarrow R_{\sigma} - 0.0005\omega^2 \quad ,$$

so the effect is one of stabilization, decreasing with frequency. The maximum value of R_{σ} obtains at $\sigma = 1$ and is 82.1. Since R_{oc} is only eight times as large, there is a good chance that this effect can be observed experimentally, for a moderately large amplitude of modulation, assuming that, at least qualitatively, these results can be extrapolated to ϵ near 1.

The value of R_{2c} for the case in which only the bottom temperature is modulated is obtained by adding the other two, so that

$$R_{2c} \rightarrow R_{\sigma} - 0.103\omega^2 \quad .$$

For σ near 1, this is not significantly different from case (b); however for larger σ , R_{σ} can become sufficiently small to be overtaken by the other terms in the sum.

As ω tends to infinity, R_{2c} tends to zero as $1/\omega^2$, so the effect of modulation disappears altogether. This agrees with Donnelly's experiments on the stability of Taylor vortices. For intermediate values of ω , the effect of changing the frequency makes itself evident in the numerator of C_n . If ω^2/σ is as large as $(n^2 + 1/2)^2\pi^4$ C_n is positive

rather than negative. Indeed, C_2 is zero when

$$\omega = \pi^2(78\sigma)^{\frac{1}{2}}/2, \quad (47)$$

so that in the symmetric case R_{2c} should be zero near that value of ω , a prediction which is borne out by the numerical calculations. Thus, for example, for $\sigma = 10$, Eq. (47) gives $\omega = 138$ while from the numerical results, R_{2c} is zero at $\omega = 145$. The peak negative value of R_{2c} is more difficult to estimate, but appears from the numerical evaluation of the series that it occurs near $\omega = 20$ and has a value of about -6 , over the entire range of σ .

None of the cases considered duplicates the behavior observed by Donnelly in his experiments, in which a peak stabilization occurs at a value of ω different from zero. This is probably due to the fact that while the two problems are fairly similar, they are not identical.

IX. ACKNOWLEDGMENTS

The author is grateful to Professor A. Ingersoll, for his valuable advice and interest in this study.

This work was supported by the Office of Naval Research.

REFERENCES

1. R.J. Donnelly, Proc. Roy. Soc., (London) A 281, 130 (1964).
2. S. Chandrasekhar, Hydrodynamic and Hydromagnetic Stability, (Oxford University Press, 1961).
3. W.V.R. Malkus and G. Veronis, J. Fluid Mech. 4, 225 (1958).
4. A. Schülter, D. Lortz and F. Busse, J. Fluid Mech. 23, (1965).
5. A. Ingersoll, Phys. Fluids 9, 682 (1966).
6. In this case it is convenient to take the wall temperature to be $T_R + \Delta T/2 + \epsilon \Delta T \cos \omega t$ at bottom and $T_R - \Delta T/2$ at the top.
7. R.E. Krishnamurti, Ph.D. Dissertation, (Univ. of California, Los Angeles 1967).

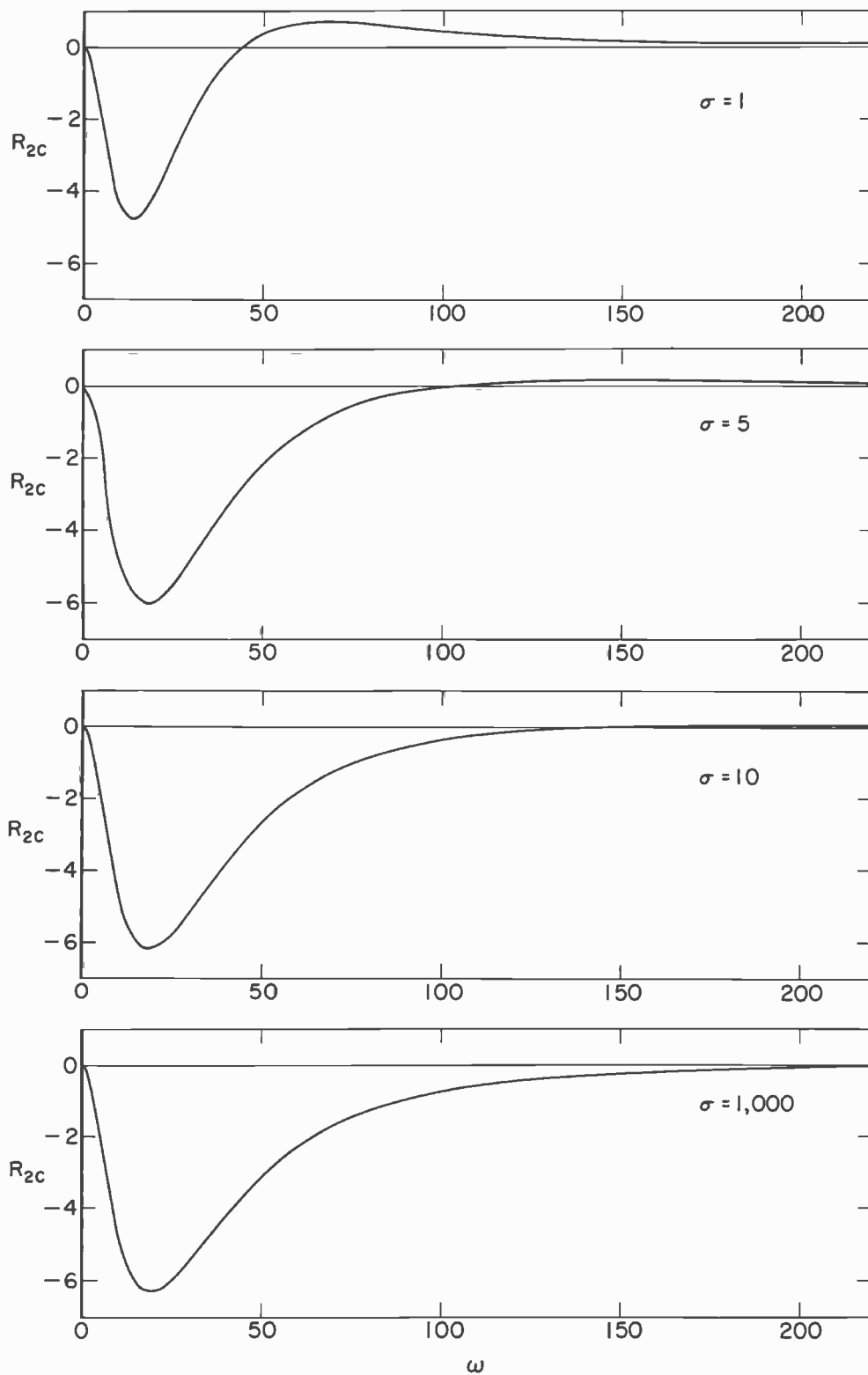


Fig. 1. R_{2c} as a function of ω when the wall temperatures are modulated in phase.

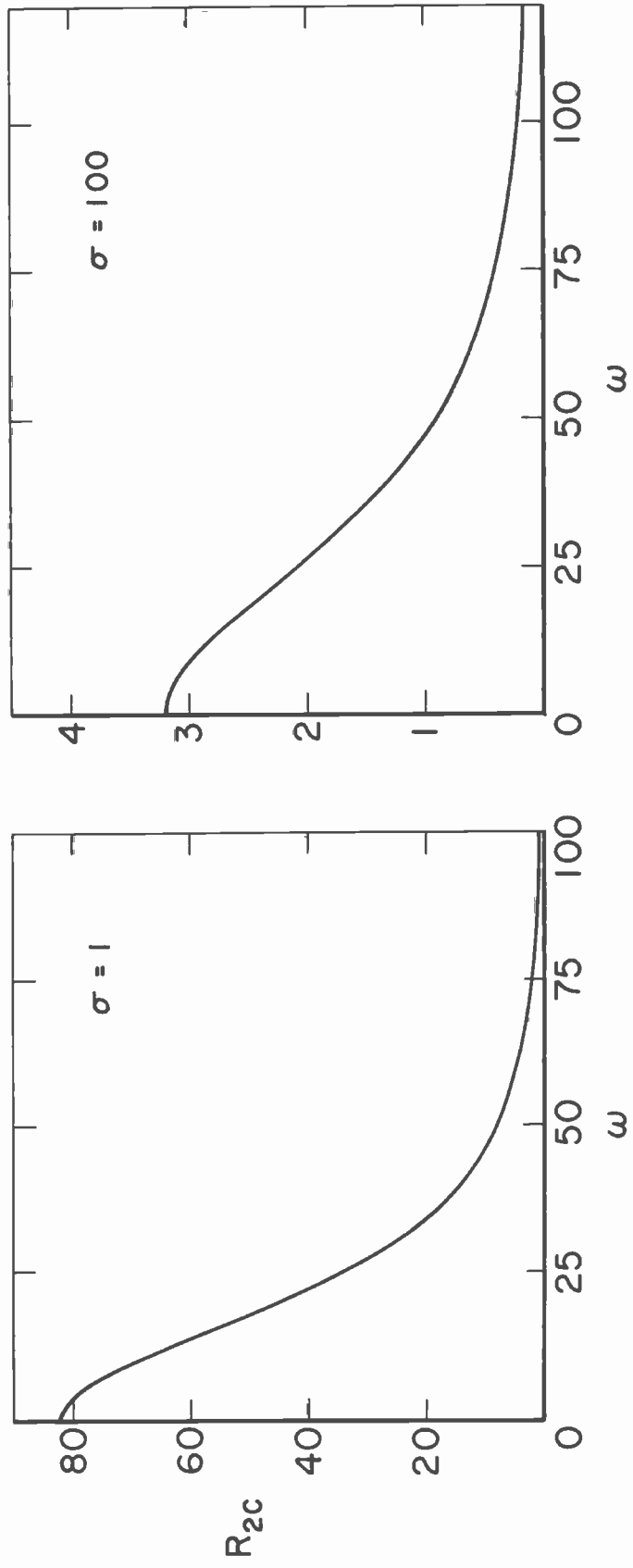


Fig. 2. R_{2c} as a function of ω when the wall temperatures are modulated out of phase. Note the change in vertical scale.

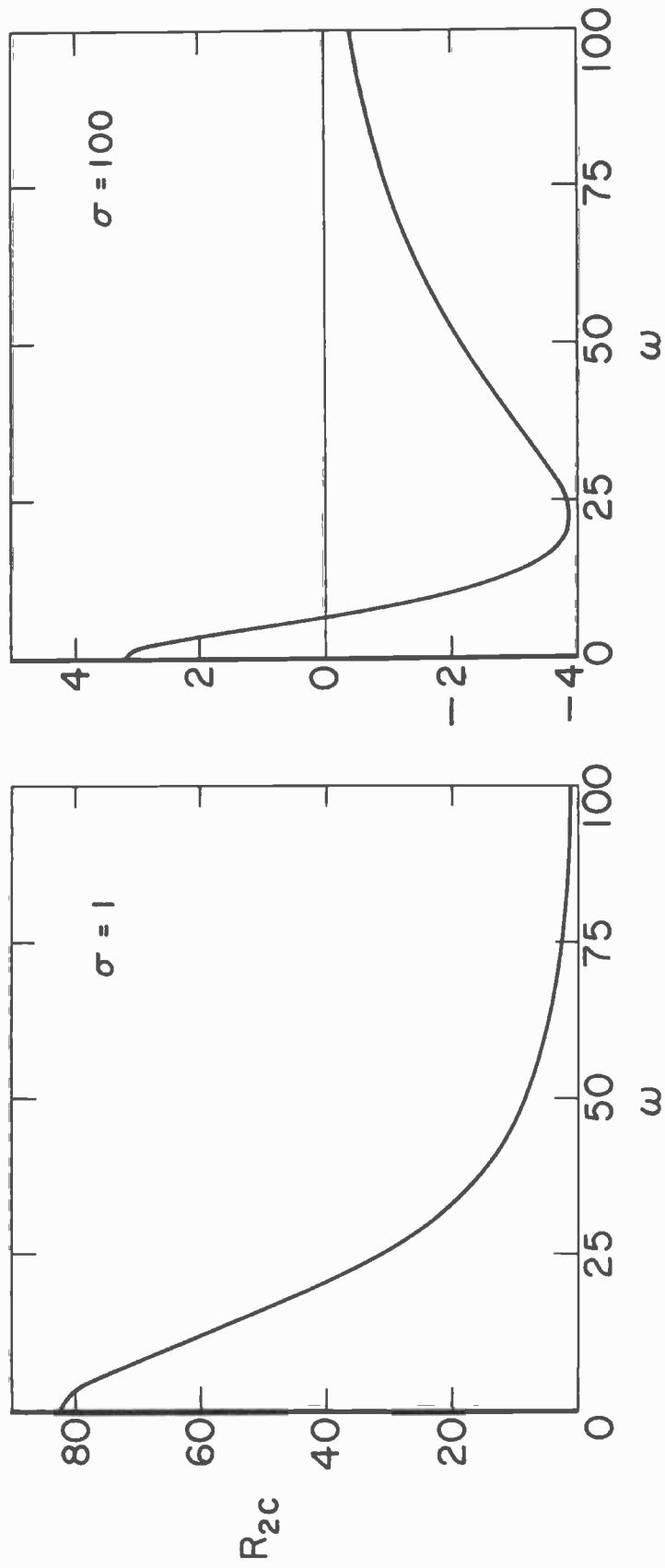


Fig. 3. R_{2c} as a function of ω when only the temperature of the lower wall is modulated. Note the change in vertical scale.

DISTRIBUTION LIS FOR UNCLASSIFIED TECHNICAL REPORTS

ISSUED UNDER

CONTRACT Nonr-220(35)

(Single copy unless otherwise specified)

Chief of Naval Research
Department of The Navy
Attn: Code 408P GHI
Washington, D. C. 20390

Office of Naval Research
San Francisco Area Office ABCDEFGHI
1076 Mission Street
San Francisco, Calif. 94103

Chief of Naval Research
Department of The Navy
Attn: Code 421 ABI
Washington, D. C. 20360

Director ABCDEFGHI
Office of Naval Research
Branch Office
219 South Dearborn St.
Chicago, Ill. 60604

Chief of Naval Research
Department of The Navy
Attn: Code 438 (3) ABCDEFGHI
Washington, D. C. 20360

Director
U. S. Naval Research Laboratory
Attn: Code 2027 ABCDEFGHI (6)
Washington, D. C. 20390

Chief of Naval Research
Department of The Navy
Attn: Code 461 ABDFGHI
Washington, D. C. 20360

Commander
Naval Ordnance Systems Command
Attn: ORD 035 ABDGI
Washington, D. C. 20360

Chief of Naval Research
Department of The Navy
Attn: Code 463 ACDEFI
Washington, D. C. 20360

Commander
Naval Ordnance Systems Command
Attn: ORD 913 (Library) ABCDEFGHI
Washington, D. C. 20360

Chief of Naval Research
Department of The Navy
Attn: Code 466 ADGHI
Washington, D. C. 20360

Commander
Naval Ordnance Laboratory
Attn: Gas Dynamics Division BCDHI
White Oak
Silver Spring, Md. 20910

Office of Naval Research
New York Area Office ABCDEFGHI
207 W. 24th St.
New York, N. Y. 10011

Commander
Naval Ordnance Laboratory
Attn: Chief, Lib. Div. ACDEFGHI
White Oak
Silver Spring, Md. 20910

Commanding Officer ABCDEFGHI
Office of Naval Research
Branch Office (25)
Box 39
FPO New York, N. Y. 09510

Commander
Naval Ordnance Laboratory
Attn: Librarian, ABCDEFGHI
White Oak
Silver Spring, Md. 20910

Director ABCDEFGHI
Office of Naval Research
Branch Office
495 Summer Street
Boston, Massachusetts 02210

Commander
Naval Weapons Laboratory
Attn: Computation & Analysis Laboratory
ACDEHI
Dahlgren, Virginia 22448

Director ABCDEFGHI
Office of Naval Research
Branch Office
1030 E. Green St.
Pasadena, Calif. 91101

Commander
Naval Weapons Laboratory
Attn: Technical Library ABCDFGHI
Dahlgren, Virginia 22418

Commander
Naval Ship Systems Command
Department of The Navy
Attn: Code 0341 CDFGHI
Washington, D. C. 20360

Commander
Naval Ship System Command
Technical Library (Code 2021) ACFI
Washington, D. C. 20360

Commander
Naval Ship Systems Command
Department of The Navy
Attn: Code 6305 DFGHI
Washington, D. C. 20360

Commander
Naval Ship Systems Command
Department of The Navy
Attn: Code 6340 ACDEFGHI
Washington, D. C. 20360

Commander
Naval Ship Systems Command
Attn: Code 6440 CDFGHI
Washington, D. C. 20360

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: Code 042 ABCDEFGHI
Washington, D. C. 20007

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: Code 108 ABCDHI
Washington, D. C. 20007

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: Code 500 CDFGHI
Washington, D. C. 20007

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: Code 513 ACDEFGHI
Washington, D. C. 20007

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: Code 520 ACDEFGHI
Washington, D. C. 20007

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: Code 521 CDFGHI
Washington, D. C. 20007

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: Code 530 DFGHI
Washington, D. C. 20007

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: Code 585 FGHI
Washington, D. C. 20007

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: J. L. Power (589) ACDFGHI
Washington, D. C. 20007

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: Code 901 ACDFGHI
Washington, D. C. 20007

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: Code 940 ACDFGHI
Washington, D. C. 20007

Commanding Officer & Director
Naval Ship Research & Development Center
Attn: Code 942 ACDFGHI
Washington, D. C. 20007

Commander
Naval Air Systems Command
Department of The Navy
Attn: Code Air 370 ACHI
Washington, D. C. 20360

Commander
Facilities Engineering Command
Department of The Navy
Attn: Code 0321 HI
Washington, D. C. 20390

Commander
Naval Ship Engineering Center
Concept Design Division
Attn: Code 6110 ACDFGHI
Washington, D. C. 20360

Commander
Naval Ship Engineering Center
Concept Design Division
Attn: Code 6420 ACDFGHI
Washington, D. C. 20360

Commander
Naval Missile Center
Attn: Technical Library ABCDEFGHI
Point Mugu, Calif. 93041

Commanding Officer & Director
Naval Civil Engineering Lab. BCDHI
Port Hueneme, Calif. 93041

Commander ACDEFGHI
Boston Naval Shipyard
Boston, Mass. 02129

Commander ACDEFGHI
Portsmouth Naval Shipyard
Portsmouth, New Hampshire 03801

Commander ACDEFGHI
Norfolk Naval Shipyard
Portsmouth, Virginia 23709

Commander ACDEFGHI
Charleston Naval Shipyard
U. S. Naval Base
Charleston, S. C. 29408

Commander
Technical Lib. (Code 249b) ACDEFGHI
Philadelphia Naval Shipyard
Philadelphia, Pa. 19112

Tech. Library, Code H245 C-3
Hunters Point Division ACDEFGHI
San Francisco Bay Naval Shipyard
San Francisco, California 94135

Shipyard Technical Library
Code 130L7 Bldg. 746 ABCDEFGHI
San Francisco Bay Naval Shipyard
Vallejo, California 94592

Superintendent
Naval Academy
Attn: Library ABCDEFGHI
Annapolis, Md. 21402

Superintendent
Naval Postgraduate School
Attn: Library ABCDEFGHI
Monterey, Calif. 93940

Librarian Station 5-2 ACDEFGI
Coast Guard Headquarters
1300 E. Street, N. W.
Washington, D. C. 20226

Office of Research & Dev. DFHI
Maritime Administration
441 G Street, N. W.
Washington, D. C. 20235

Division of Ship Design CDFGI
Maritime Administration
441 G Street, N. W.
Washington, D. C. 20235

Department of The Army
Coastal Engrg. Research Center I
5201 Little Falls Road, N. W.
Washington, D. C. 20011

Redstone Scientific Information Center
Attn: Chief, Document Section ABCDEFGHI
Army Missile Command
Redstone Arsenal, Alabama 35809

Dr. J. Menkes ABCDEFGHI
Institute for Defense Analyses
400 Army-Navy Drive
Arlington, Virginia 22204

NASA, Langley Research Center
Langley Station
Attn: Library MS 185 ABCDEFGHI
Hampton, Virginia 23365

NASA, Lewis Research Center
Attn: Library MS 60-3 ABCDEFGI
21000 Brookpark Road
Cleveland, Ohio 44135

NASA Scientific & Tech. Info. Fac. ABCDEFGHI
Attn: Acquisitions BR (S-AK/DL)
P. O. Box 33
College Park, Maryland 20740

Science & Technology Div. ABCDEFGHI
Library of Congress
Washington, D. C. 20540

National Academy of Sciences
National Research Council ABCDEFGHI
2101 Constitution Ave., N. W.
Washington, D. C. 20360

Director
Engrg. Science Division ACDFGHI
National Science Foundation
Washington, D. C. 20550

National Science Foundation
Engineering Division ABCDEI
1800 G Street, N. W.
Washington, D. C. 20550

Director
National Bureau of Stds. I
Attn: Dr. G. B. Schubauer
Washington, D. C. 20234

Defense Documentation Ctr. ABCDEFGHI
Cameron Station (20)
Alexandria, Virginia 22314

Society of Naval Architects and
Marine Engineers CDFGI
74 Trinity Place
New York, N. Y. 10006

Engineering Societies Library ABCDEFGHI
345 East 47th Street
New York, N. Y. 10017

Professor M. S. Plesset
Calif. Inst. of Technology
Pasadena, California 91109

Professor A Acosta ACDFGHI
Calif. Inst. of Technology
Pasadena, California 91109

Professor E. E. Sechler ABCDEFGHI
Executive Officer for Aero.
Calif. Inst. of Technology
Pasadena, California 91109

Professor J. V. Wehausen CFHI
Department of Naval Architecture
University of California
Berkeley, California 94720

Professor J. R. Paulling DFGHI
Department of Naval Architecture
University of California
Berkeley, California 94720

Dr. L. Talbot ABI
University of California
Department of Engineering
Berkeley, California 94720

Professor E. V. Laitone BFGHI
University of California
Berkeley, California 94720

Professor J. Johnson ADHI
412 Hesse Hall
University of California
Berkeley, California 94720

Librarian ACDEFGHI
Department of Naval Architecture
University of California
Berkeley, California 94720

Director ACDEFGHI
Scripps Institution of Oceanography
University of California
La Jolla, California 92037

Professor John Miles CFI
% I. G. P. P.
University of Calif. San Diego
La Jolla, Calif. 92038

Professor R. W. Leonard ADI
University of California
Los Angeles, California 90024

Professor John Laufer ACDHI
Dept. of Aerospace Engineering
University of Southern California
Los Angeles, California 90007

Professor R. C. MacCamy FI
Department of Mathematics
Carnegie Institute of Technology
Pittsburgh, Pennsylvania 15213

Dr. Martin H. Bloom ABCDFI
Polytechnic Inst. of Brooklyn
Graduate Center
Dept. of Aerospace
Eng. & Applied Mechanics
Farmingdale, N. Y. 11735

Professor Maurice L. Albertson FI
Department of Civil Engineering
Colorado State University
Fort Collins, Colorado 80521

Professor E. L. Resler ABCDHI
Graduate School of Aeronautical Engrg.
Cornell University
Ithaca, New York 14851

Professor G. H. Carrier HI
Harvard University
Cambridge, Massachusetts 02138

Professor G. Birkhoff ACEFI
Harvard University
Cambridge, Massachusetts 02138

Dr. Charles Dalton DI
University of Houston
Dept. of Mechanical Engineering
Houston, Texas 77004

Dr. Clark Goodman, Chairman EHI
Physics Department
University of Houston
3801 Cullen Boulevard
Houston, Texas 77004

School of Applied Mathematics
Indiana University ABCDEFGHI
Bloomington, Indiana 47401

Prof. J. F. Kennedy, Director ACDEFGHI
Iowa Institute of Hydraulic Research
State University of Iowa
Iowa City, Iowa 52240

Professor L. Landweber ACDFGHI
Iowa Institute of Hydraulic Research
State University of Iowa
Iowa City, Iowa 52240

Professor O. M. Phillips AHI
The Johns Hopkins University
Baltimore, Maryland 20910

Professor M. V. Morkovin ACGI
Aeronautics Building
Johns Hopkins University
Baltimore, Maryland 21218

Professor Pai ABI
Inst. for Fluid Dynamics and
Applied Mathematics
University of Maryland
College Park, Maryland 20742

Professor A. T. Ippen ACDFGHI
Massachusetts Inst. Of Tech.
Cambridge, Massachusetts 02139

Professor M. A. Abkowitz ACDFGHI
Dept. of Naval Architecture and
Marine Engineering
Massachusetts Inst. of Tech.
Cambridge, Massachusetts 02139

Department of Naval Architecture &
Marine Engineering ACDFGHI
Room 5-228
Massachusetts Inst. of Tech.
Cambridge, Massachusetts 02139

Professor R. F. Probststein ABCDFHI
Department of Mechanical Engrg.
Massachusetts Inst. of Tech.
Cambridge, Massachusetts 02139

Professor P. Mandel DFGHI
Room 5-325
Massachusetts Inst. Tech.
Cambridge, Massachusetts 02139

Professor E. Mollo-Christensen AHI
Room 54-1722
Massachusetts Inst. of Technology
Cambridge, Massachusetts 02139

Professor Finn C. Michelsen CDFGI
Naval Architecture & Marine Engrg.
450 West Engineering Building
University of Michigan
Ann Arbor, Michigan 48108

Professor W. W. Willmarth ACDFHI
Dept. of Aero/Space Engineering
University of Michigan
Ann Arbor, Michigan 48108

Professor F. G. Hammitt
University of Michigan ACDEFGI
Ann Arbor, Michigan 48108

Dr. C. S. Yih CDFI
Department of Engineering Mechanics
University of Michigan
Ann Arbor, Michigan 48108

Mr. J. M. Wetzel DCFGHI
St. Anthony Falls Hydraulic Lab.
University of Minnesota
Minneapolis, Minnesota 55414

Lorenz G. Straub Library ACDFGHI
St. Anthony Falls Hydraulic Lab.
Mississippi River at 3rd Ave., S. E.
Minneapolis, Minnesota 55414

C. E. Bowers (2) CDFHI
St. Anthony Falls Hydraulic Lab.
University of Minnesota
Minneapolis, Minnesota 55414

Mr. C. S. Song CDFGI
St. Anthony Falls Hydraulic Lab.
University of Minnesota
Minneapolis, Minnesota 55414

Professor A. Peters CFHI
Institute of Mathematical Sciences
New York University
251 Mercer Street
New York, N. Y. 10003

Professor J. J. Stoker AFGHI
Institute of Mathematical Sciences
New York University
251 Mercer Street
New York, New York 10003

Professor W. J. Pierson, Jr. AHI
New York University
Dept. of Meteorology & Oceano.
University Heights
New York, New York 10405

Professor J. J. Foody CDI
Chairman, Engineering Department
State University of New York
Maritime College
Bronx, New York 10465

Professor Ali Bulent Cambel ADFI
Department of Mechanical Engineering
Northwestern University
Evanston, Illinois 60201

Professor A. Charnes ACDEFGHI
The Tech. Institute
Northwestern University
Evanston, Illinois 60201

Professor A. G. Strandhagen ACDFGI
Department of Engineering Mechanics
University of Notre Dame
Notre Dame, Indiana 46556

Dr. M. Sevik ACDFGHI
Ordnance Research Laboratory
Pennsylvania State University
University Park, Pennsylvania 16801

Professor Allen Chapmann, Chairman BHI
Mechanical Engineering Dept.
William M. Rice Institute
Box 1892
Houston, Texas 77001

Dr. Byrne Perry CFHI
Department of Civil Engineering
Stanford University
Stanford, California 94305

Professor E. Y. Hsu CDFHI
Dept. of Civil Engineering
Stanford University
Stanford, California 94305

Dr. J. P. Breslin ACDFGHI
Stevens Institute of Technology
Davidson Laboratory
Hoboken, New Jersey 07030

Mr. D. Savitsky ACDFGHI
Stevens Institute of Technology
Davidson Laboratory
Hoboken, New Jersey 07030

Professor E. V. Lewis DGI
Webb Institute of Naval Architecture
Glen Cove, Long Island, N. Y. 11542

Technical Library ACDFGHI
Webb Institute of Naval Architecture
Glen Cove, Long Island, New York 11542

National Research Council ABCDEFGHI
Aeronautical Library
Attn: Miss. O. M. Leach, Librarian
Montreal Road
Ottawa 7, Canada

Prof. Dr. G. P. Weinblum, Dir. FI
Institut für Schiffbau
Universität der Hamburg
Lammersbeth 90
2 Hamburg 33, Germany

Dr. O. Grim ACDI
Institute für Schiffbau
Lammersbeth 90
2 Hamburg 33, Germany

Dr. H. W. Lerbs ACDFI
Hamburgische Schiffbauversuchsanstalt
Bramfelder Strasse 164
Hamburg 33, Germany

Dr. K. Hasselman AHI
Institut für Schiffbau der
Universität Hamburg
Lammersbeth 90
2 Hamburg 33, Germany

Dr. K. Eggers ACDFGI
Institut für Schiffbau
Universität der Hamburg
Lammersbeth 90
2 Hamburg 33, Germany

Dr. H. Reichardt, Dir. ACDFGI
Mas Planck Institut für Stromungsforschung
Bottingerstrasse 6-8
Gottingen, Germany

Mr. A. Silverleaf CDFI
National Physical Laboratory
Teddington, Middlesex, England

Manchester University I
Manchester, England
Attn: Professor F. Ursell

Mr. C. Wigley FI
Flat 103
6-9 Charterhouse Square
London E, C, 1, England

Prof. Ir. J. Gerritsma ACDFGHI
Head Shipbuilding Lab., Tech. Univ.
Mekelweg 2
Delft, The Netherlands

Ir. W. Spuyman ACDEFGHI
Netherlands Ship Research Centre
Mekelweg 2
Delft, The Netherlands

Prof. Dr. Ir. J. D. Van Manen ACDFGI
Netherlands Ship Model Basin
Haagsteeg 2, Postbox 28
Wageningen, The Netherlands

Professor J. K. Lunde ACDEFGI
Skipmodelltanken
Trondheim, Norway

Professor Carl Prohaska ACDFGI
Hydro-og Aerodynamisk Laboratorium
Lyngby, Denmark

Professor S. Siestrunck CFI
Bureau D'Analyse de Recherches
Appliquees
6 Rue Louis Pasteur
92 Boulogne, France

Dr. C. Elata DI
Hydraulics Laboratory
Israel Institute of Technology
Haifa, Israel

Defence Research & Dev. Attache
ACDEFGI
Australian Embassy
1735 Eye Street, N. W.
Washington, D. C. 20006

Mr. Alfonso Alcedan L., Dir. ACDEFGI
Laboratorio Nacional de Hydraulics
Antiguo Cameno A. Ancon
Casilla Jostal 682
Lima, Peru

C. A. Gongwer ACDFGHI
Aerojet General Corporation
9100 E. Flair Drive
El Monte, California 91734

Library ABCDEFGHI
Aerojet-General Corp.
6352 N. Irwindale Ave.
Azusa, California 91702

Dr. Sandri AI
Aeronautical Research Assoc. of
Princeton
50 Washington Road
Princeton, New Jersey 08540

Myron J. Block, President AHI
Block Engineering, Inc.
19 Blackstone Street
Cambridge, Massachusetts 02139

Dr. Irving C. Statler, Head CFI
Applied Mechanics Department
Cornell Aeronautical Laboratory Inc.
P. O. Box 235
Buffalo, New York 14221

Mr. F. Dell' Amico ADFGI
Cornell Aeronautical Laboratory
Buffalo, New York 14221

Dr. R. B. Couch CDFGHI
General Dynamics Corp.
Quincy Division
97 E. Howard
Quincy, Massachusetts 02169

Dr. S. F. Hoerner CDFGI
148 Busteed Drive
Midland Park, New Jersey 07432

Director ACDGHI
Hudson Laboratories
Dobbs Ferry, New York 10522

Mr. P. Eisenberg, President ACDEFGHI
Hydronautics, Inc.
Pindell School Rd.
Howard County
Laurel, Md. 20810

Dr. H. Cohen AFHI
IBM Research Center
P. O. Box 218
Yorktown Heights, N. Y. 10598

Mr. Richard Barakat AFHI
Optics Department
Itek Corporation
Lexington, Massachusetts 02173

Mr. R. W. Kermeen ACDFGHI
Lockheed Missiles & Space Company
Department 57101 Bldg. 150
Sunnyvale, California 94086

Engineering Library ABCDEFGHI
Dept. 218, Bldg. 101
McDonnell Aircraft Corp.
P. O. Box 516
St. Louis, Missouri 63166

Mr. Schuyler Kleinhans ACDEFGI
Vice President - Engineering
McDonnell-Douglas Aircraft Co.
Santa Monica, California 90406

Dr. Paul Kaplan ACDFGHI
Oceanics, Inc.
Plainview, Long Island, N. Y. 11803

Dr. Alfred Ritter ABGHI
Therm Advanced Research, Inc.
100 Hudson Circle
Ithaca, New York 14850

Dr. Jack Kotik ACDFGHI
Trg. Incorporated
Route 110
Melville, New York 11746

Editor ABCDEFGHI
Applied Mechanics Review
Southwest Research Institute
8500 Culebra Road
San Antonio, Texas 78206

Dr. H. N. Abramson ABCFGHI
Southwest Research Institute
8600 Culebra Road
San Antonio, Texas 78228

Mr. W. T. Hamilton ABDGI
Chief - SST Technology
Supersonic Transport Division
P.O. Box 733
Renton, Washington 98055

Dr. F. W. Boggs ACDFGHI
U. S. Rubber Company
Research Center
Wayne, New Jersey 07470

DOCUMENT CONTROL DATA - R&D

(Security classification of title, body of abstract and indexing annotation must be entered when the overall report is classified)

| | | | |
|--|--|--|--|
| 1. ORIGINATING ACTIVITY <i>(Corporate author)</i> California Institute of Technology Division of Engineering and Applied Science | | 2a. REPORT SECURITY CLASSIFICATION Unclassified | |
| | | 2b. GROUP Not Applicable | |
| 3. REPORT TITLE EFFECT OF MODULATION ON THE ONSET OF THERMAL CONVECTION | | | |
| 4. DESCRIPTIVE NOTES <i>(Type of report and inclusive dates)</i> Technical Report | | | |
| 5. AUTHOR(S) <i>(Last name, first name, initial)</i> Venezian, Giulio | | | |
| 6. REPORT DATE March 1968 | 7a. TOTAL NO. OF PAGES 17 | 7b. NO. OF REFS. 7 | |
| 8a. CONTRACT OR GRANT NO. Nonr 220(35) | 9a. ORIGINATOR'S REPORT NUMBER(S) Report No. 97-10 | | |
| b. PROJECT NO. c. d. | 9b. OTHER REPORT NO(S) <i>(Any other numbers that may be assigned this report)</i> | | |
| 10. AVAILABILITY/LIMITATION NOTICES Distribution of this document is unlimited. | | | |
| 11. SUPPLEMENTARY NOTES | | 12. SPONSORING MILITARY ACTIVITY Office of Naval Research | |
| 13. ABSTRACT The stability of a horizontal layer of fluid heated from below is examined when, in addition to a steady temperature difference between the walls of the layer, a time-dependent sinusoidal perturbation is applied to the wall temperatures. Only infinitesimal disturbances are considered. The effects of the oscillating temperature field are treated by a perturbation expansion in powers of the amplitude of the applied field. The shift in the critical Rayleigh number is calculated as a function of frequency, and it is found that it is possible to advance or delay the onset of convection by time modulation of the wall temperatures. | | | |

| 14. KEY WORDS | LINK A | | LINK B | | LINK C | |
|--|--------|----|--------|----|--------|----|
| | ROLE | WT | ROLE | WT | ROLE | WT |
| Thermal convection Benard convection Stability of heated fluid layer | | | | | | |

INSTRUCTIONS

1. **ORIGINATING ACTIVITY:** Enter the name and address of the contractor, subcontractor, grantee, Department of Defense activity or other organization (*corporate author*) issuing the report.

2a. **REPORT SECURITY CLASSIFICATION:** Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Marking is to be in accordance with appropriate security regulations.

2b. **GROUP:** Automatic downgrading is specified in DoD Directive 5200.10 and Armed Forces Industrial Manual. Enter the group number. Also, when applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.

3. **REPORT TITLE:** Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capitals in parenthesis immediately following the title.

4. **DESCRIPTIVE NOTES:** If appropriate, enter the type of report, e.g., interim, progress, summary, annual, or final. Give the inclusive dates when a specific reporting period is covered.

5. **AUTHOR(S):** Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, show rank and branch of service. The name of the principal author is an absolute minimum requirement.

6. **REPORT DATE:** Enter the date of the report as day, month, year; or month, year. If more than one date appears on the report, use date of publication.

7a. **TOTAL NUMBER OF PAGES:** The total page count should follow normal pagination procedures, i.e., enter the number of pages containing information.

7b. **NUMBER OF REFERENCES:** Enter the total number of references cited in the report.

8a. **CONTRACT OR GRANT NUMBER:** If appropriate, enter the applicable number of the contract or grant under which the report was written.

8b, 8c, & 8d. **PROJECT NUMBER:** Enter the appropriate military department identification, such as project number, subproject number, system numbers, task number, etc.

9a. **ORIGINATOR'S REPORT NUMBER(S):** Enter the official report number by which the document will be identified and controlled by the originating activity. This number must be unique to this report.

9b. **OTHER REPORT NUMBER(S):** If the report has been assigned any other report numbers (*either by the originator or by the sponsor*), also enter this number(s).

10. **AVAILABILITY/LIMITATION NOTICES:** Enter any limitations on further dissemination of the report, other than those

imposed by security classification, using standard statements such as:

- (1) "Qualified requesters may obtain copies of this report from DDC."
- (2) "Foreign announcement and dissemination of this report by DDC is not authorized."
- (3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through _____."
- (4) "U. S. military agencies may obtain copies of this report directly from DDC. Other qualified users shall request through _____."
- (5) "All distribution of this report is controlled. Qualified DDC users shall request through _____."

If the report has been furnished to the Office of Technical Services, Department of Commerce, for sale to the public, indicate this fact and enter the price, if known.

11. **SUPPLEMENTARY NOTES:** Use for additional explanatory notes.

12. **SPONSORING MILITARY ACTIVITY:** Enter the name of the departmental project office or laboratory sponsoring (*paying for*) the research and development. Include address.

13. **ABSTRACT:** Enter an abstract giving a brief and factual summary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the information in the paragraph, represented as (TS), (S), (C), or (U).

There is no limitation on the length of the abstract. However, the suggested length is from 150 to 225 words.

14. **KEY WORDS:** Key words are technically meaningful terms or short phrases that characterize a report and may be used as index entries for cataloging the report. Key words must be selected so that no security classification is required. Identifiers, such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical content. The assignment of links, roles, and weights is optional.